# Adaptive Load Balance Techniques in Parallel Rarefied Gas Simulations

S. Antonov,\* F.-J. Pfreundt,\* and J. Struckmeier†

\*Institute for Industrial Mathematics, Kaiserslautern, Germany; and †Department of Mathematics, University of Kaiserslautern, Germany E-mail: struckm@mathematik.uni-kl.de

Received July 26, 1996; revised December 30, 1996

The paper presents some adaptive load balance techniques for the simulation of rarefied gas flows on parallel computers. It is shown that a static load balance is insufficient to obtain a scalable parallel efficiency. Hence, two adaptive techniques are investigated which are based on simple algorithms. Numerical results show that using heuristic techniques one can achieve a sufficiently high efficiency over a wide range of different hardware platforms. © 1997 Academic Press

*Key Words:* rarefied gas flows; particle methods; parallel computing; adaptive load balance.

# **1. INTRODUCTION**

Particle or Monte Carlo methods are efficient numerical tools to predict rarefied gas flows around reentry bodies. Besides the well-known DSMC approach developed by Bird [7], the Finite-Pointset Method as described in [11] has been widely used to investigate rarefied gas flows [5, 6, 2]. Moreover, it is known, that, in general, Monte Carlo methods can be implemented more easily on parallel architecture than other methods, like, e.g., FEM-computations. The same holds for particle schemes for rarefied gas simulations and several authors already investigated parallel versions of the classical serial codes (see [3, 4, 10, 12, 14, 17] and the references given there). To achieve an appropriate parallel efficiency, it is necessary to introduce adaptive load balance concepts. The main reason for this is the strong variation of the macroscopic density within the spatial domain. In the current paper we present some load balance concepts which result in very efficient parallel particle schemes for rarefied gas flows. The paper is organized as follows: in Section 2 we briefly describe the mathematical equation which describes rarefied gas flows and explain the main ideas how to simulate this equation using particle methods. Moreover, we indicate how to implement those schemes on a parallel architecture and why an adaptive load balance becomes necessary to obtain scalable parallel schemes. Section 3 deals with some heuristic methods to obtain a sufficiently accurate load balance and in particular we focus on two different approaches: the so-called Min–Max-Update as well as a Streamline Method. Heuristic methods are necessary due to the high dimensionality of the underlying optimization problem. The following section is devoted to a wide range of applications and includes a comparison of various hardware platforms. A conclusion of the current paper is given in Section 5.

#### 2. RAREFIED GAS SIMULATION BY PARTICLE METHODS

# 2.1. Particle Methods

Rarefied gas flows are described by the well-known Boltzmann equation introduced by Ludwig Boltzmann in 1872, a nonlinear transport equation which defines the time evolution of the density function of the gas in the phase space. In this section we consider the Boltzmann equation on the spatial domain  $\Omega \subset \mathbb{R}^3$  defined as the nonlinear transport equation

$$\frac{\partial f}{\partial t} + v \cdot \nabla_x f = \frac{1}{\varepsilon} Q(f)$$

for the density function f = f(t, x, v) of the gas, together with the initial condition

$$f(0, x, v) = f(x, v)$$

and appropriate boundary conditions on  $\partial \Omega$ , like inflow and outflow conditions or some scattering conditions of the form

$$|(v,n)|f(t,x,v) = \int_{(v',n)<0} R(v,v',t,x)|(v',n)|f(t,x,v') \, \mathrm{d}v', \qquad (v,n) > 0,$$

where  $R(v \rightarrow v'; t, x)$  is an appropriate gas-surface scattering kernel. The collision operator, here written down for the simplest case of a monoatomic gas, is expressed in the form

$$Q(f) = \int_{\mathbb{R}^3} \int_{S^2_+} k(|v - v_*|, n) \{ f'f'_* - ff_* \} \, \mathrm{d}n \, \mathrm{d}v_*,$$

where we used the notation

$$f' = f(t, x, v'), \qquad v' = v - (v - v_*, n)n$$
  
$$f'_* = f(t, x, v'_*), \qquad v'_* = v_* + (v - v_*, n)n.$$

Besides the simple monoatomic case, one may consider a rarefied gas mixture consisting of M different species, where the different species might carry internal energies and the whole system might undergo chemical reactions, like dissociation–recombination reactions [13].

*Remark* 2.1. It turns out that, concerning parallelization, the particle schemes for monoatomic gas and gas mixtures with chemical reactions are very similar. Hence, we will restrict ourselves in the first part of the paper to the monoatomic Boltzmann equation. Nevertheless, the results presented in Section 4 include computations with real gas effects.

The most efficient numerical tools to simulate the Boltzmann equation are Particle Methods (also called Monte Carlo Methods), where the solution f = f(t, x, v) is approximated at every time  $t \ge 0$  by a finite set of particles in the phase space  $\Omega \times \mathbb{R}^3$ . A finite set of particles is defined as the set { $(\alpha_1, x_1, v_1), ..., (\alpha_n, x_n, v_n)$ }, where  $(x_i, v_i) \in \Omega \times \mathbb{R}^3$ , i = 1, ..., n, and  $\alpha_i$ , i = 1, ..., n are the weights of the particles. Both particle characteristics, i.e., points in the phase space as well as the weights, might depend on time

$$(x_i, v_i) = (x_i(t), v_i(t)), \qquad \alpha_i = \alpha_i(t), i = 1, ..., n$$

and the principle of a particle scheme is to derive a time evolution on the level of the discrete approximation by particles. It is beyond the scope of the paper to describe in detail the derivation of such particle schemes. Hence, we restrict ourselves to the main steps of the particle method:

To derive the time evolution of the particles, one uses a splitting method based on a discrete time step  $\Delta t$ , i.e., over the time interval  $[n \ \Delta t, (n + 1) \ \Delta t)$  one solves successively the two equations

$$\frac{\partial f}{\partial t} + v\nabla_x f = 0 \tag{2.1}$$

and

$$\frac{\partial f}{\partial t} = Q(f). \tag{2.2}$$

1. The Free flow of particles. Equation (2.1) can be solved by the method of characteristics, which leads to the simple free flow condition on the level of the particle approximation, i.e.,

$$x_i((n+1)\Delta t) = x_i(n\Delta t) + \Delta t v_i(n\Delta t), \quad v_i((n+1)\Delta t) = v_i(n\Delta t).$$
(2.3)

If the particle trajectory defined by (2.3) intersects with the boundary  $\delta\Omega$ , one has to consider the corresponding boundary condition.

2. Collision process between particles. To obtain the particle scheme for Eq. (2.2) one replaces the exact collision operator Q(f) by a spatially smoothed operator

 $Q^{\Delta x}(f)$ . Based on a partition  $\bigcup_{k \in \mathcal{X}} Z_k^{\Delta x} = \Omega$  of the spatial domain, such that diam  $Z_k^{\Delta x} \leq \Delta x$ , and a smoothing kernel  $\beta^{\Delta x}$  given by

$$\beta^{\Delta x}(x, x_*) = \sum_{k \in \mathcal{X}} \frac{\mathscr{X}_{Z_k}^{\Delta x}(x) \mathscr{X}_{Z_k}^{\Delta x}(x_*)}{\operatorname{Vol}(Z_k^{\Delta x})}$$

the collision operator Q(f) is replaced by

$$Q^{\Delta x}(f) = \int_{\Omega} \int_{\mathbb{R}^3} \int_{S^2_+} \beta^{\Delta x}(x, x_*) k(|v - v_*|, n) \{f'f'_* - ff_*\} \, \mathrm{d}n \, \mathrm{d}v_* \, \mathrm{d}x_*$$

with the notations

$$f' = f(t, x, v'), \qquad f'_* = f(t, x_*, v'_*),$$
etc.

Due to the special form of the smoothing kernel  $\beta^{\Delta x}$ , the problem is reduced to a system of spatial-homogeneous Boltzmann equations on the cell system  $Z_k^{\Delta x}$  and one continues by applying some discretization techniques for the spatial-homogeneous Boltzmann equation. In particular the spatial coordinate of each particle remains constant during this collision step.

## 2.2. Particle Methods on Parallel Computers

A parallel version of particle methods for rarefied gas simulations is obtained by dividing the spatial domain  $\Omega$  into k nonoverlapping subdomains, where k is the number of processors. Then, each processor computes the time evolution of those particles belonging to its own subdomain. Communication routines between processors handle the particles which leave subdomains during the free flow condition, the first part of the splitting approach, whereas no communication is necessary during the second step, the collisions between particles, because the spatial coordinates remain constant.

The following two criteria should be used to obtain an appropriate partition on the processors:

• each processor should use nearly the same CPU-time to compute the time evolution of the particles belonging to the processor,

• the communication between the processors should be as small as possible.

Because rarefied flow simulations are time-dependent simulations, it is in general not possible to fulfill both criteria using a static partition of the computational domain. The following results (Table I) are obtained on an nCUBE 2S parallel system using a static partition for the rarefied gas flow around the EXPRESS reentry capsule (see Sect. 4.1).

*Remark* 2.2. Concerning the definition of the speedup factor, the efficiency, and the load balance coefficient (lbc), we refer the reader to Section 4, where we present detailed numerical investigations.

	Parallel Efficie	ency with Static	Load Balance		
Proc.	CPU [m:s]	Speedup	Effic.	lbc	
64	10:13	23.2	36.3	2.61	
32	11:15	21.1	65.9	1.48	
16	21:25	11.1	69.2	1.45	
8	35:20	6.7	83.9	1.20	
4	67:19	3.5	88.1	1.16	

 TABLE I

 Parallel Efficiency with Static Load Balance

Due to the shock front around the reentry body, the density field is strongly inhomogeneous and this results in an insufficient load balance. A priori information on the regions with high density is not accurate enough to obtain a priori partitions which overcome this effect. A further reason is that in transient computations the number of simulation particles does not remain constant, but increases due to the formation of the shock front. Moreover, in solving larger problems a static partition may result in a memory overflow on certain processors. Hence, adaptive load balance techniques are absolutely necessary to obtain efficient parallel codes.

# **3. ADAPTIVE LOAD BALANCE TECHNIQUES**

As mentioned in the previous section, it is necessary to use adaptive load balance concepts to obtain efficient particle schemes which are scalable with respect to the number of processors as well as the flow characteristics. In the case of rarefied flow simulations, an adaptive load balance technique might be implemented by changing the partition of subdomains on the processors within the transient computation. To perform this partition it is appropriate to introduce the so-called Load Balance Units (lbu's) which are defined as the smallest subdomains that may be exchanged between processors. Here, the lbu's consist of a certain number of cells  $Z_k^{\Delta x}$  used to perform the collision process between particles. This concept was already used in [14], where the load balance units consist of sticks along the flow direction.

*Remark* 3.1. The partition updating of the domain should be implemented in the particle scheme after the first step of the splitting approach. Due to the partition updating a communication step between the processors becomes necessary and this can be done in combination with the exchange of particles due to the free flow condition.

With the concept of lbu's, the partition updating problem may be formalized as follows: Let k be the number of processors  $\{p_1, ..., p_k\}$ , p the number of lbu's  $\{g_1, ..., g_p\}$ . Then a partition of lbu's on the set of processors is given by a mapping  $P : \{1, ..., p\} \rightarrow \{1, ..., k\}$ , such that the lbu  $g_i$  belongs to processor  $p_j$ , if P(i) = j. Moreover, let  $t_i$  be the CPU-time used to compute to solution on lbu  $g_i$ . Then, the CPU-time  $t^j$  on processor  $p_j$  is given by the formula

$$t^j = \sum_{i=1}^p t_i \,\delta_{jP(i)},$$

where

$$\delta_{kl} = \begin{cases} 0 & \text{if } k \neq l \\ 1 & \text{if not} \end{cases}$$

and the averaged CPU-time  $t_{av}$  is defined as

$$t_{\rm av} = \frac{1}{k} \sum_{i=1}^{k} t^i.$$

Now, let  $\mathscr{P}$  be the set of all mappings  $P : \{1, ..., p\} \to \{1, ..., k\}$  and for given  $\varepsilon > 0$  we define the set  $\mathscr{P}_{\varepsilon}$  of all  $P \in \mathscr{P}$  such that

$$\max_{i=1,\ldots,k}|t_{\mathrm{av}}-t^i|<\varepsilon.$$

First of all, partitions belonging to  $\mathscr{P}_{\varepsilon}$  ensure an appropriate load balance with respect to the parameter  $\varepsilon$  and if we perform a partition updating of the spatial domain, we should choose an element out of  $\mathscr{P}_{\varepsilon}$ . Moreover, we like to choose a partition which results in a low communication between the processors. This is formalized by considering the flux  $F_{ij}$  of particles from lbu  $g_i$  to  $g_j$ . Then, the total flux of particles between the processors based on the partition  $P \in \mathscr{P}$  is given by

$$F^{\mathcal{P}} = \sum_{i=1}^{k} \sum_{j,l=1}^{p} F_{ij} \delta_{iP(j)} (1 - \delta_{iP(l)}).$$

If we denote by P' the present partition of  $\Omega$ , then it is moreover necessary to consider the communication effort to pass from P' to P. This cost  $K_{P'\to P}$  can be expressed by

$$K_{P' \rightarrow P} = \sum_{i=1}^{k} \sum_{j=1}^{p} \delta_{iP'(j)} (1 - \delta_{iP(j)}).$$

If we denote  $\mathring{K}$  the bound that we want to impose on the costs  $K_{P' \to P}$ , we finally obtain the following minimization problem:

For given  $\varepsilon$ ,  $\mathring{K}$ , and partition P', find the minimizer on the set  $\mathscr{P}_{\varepsilon}$  of the functional  $F_P$ , under the restriction that  $K_{P' \to P} \leq \mathring{K}$ .

*Remark* 3.2. Instead of using the CPU-time of each processor, one might use some estimates on the CPU-times, like the number of particles each processor has to handle. This estimate turns out to be accurate as long as the gas-surface interaction requires only a small part of the total CPU-time, e.g., if the collisions between particles are the dominating part. It is obvious that this is a difficult optimization problem and, moreover, it is not clear if there exists a solution for arbitrary values of  $\varepsilon$  and  $\hat{K}$ .

Hence, assuming that there exists an optimal solution, one has to use some heuristic algorithms to obtain a computationally cheap solution which is close to the optimal solution. Two heuristic concepts are described in the following:

• The Min-Max-Update. A similar technique was already used in [14], based on the number of particles on each processor.

• The Streamline Approach. This technique uses the concept of streamlines in order to reduce the communication between the processors.

# 3.1. Min-Max-Update

The Min-Max-Update is a simple idea used to obtain a partition updating using the present partition P, but it does not consider the effort for the communication between processors. Some lbu's are simply moved to other processors, such that the present partition P' becomes an element of  $\mathscr{P}_{\varepsilon}$  for some small  $\varepsilon > 0$ . This algorithm is described as follows:

MIN-MAX-UPDATE. (1) Let *P* be the present partition and define for each processor *a* switch  $s_i \in \{0, 1\}$ , by

$$s_i = \begin{cases} 0 & \text{if } \sum_{j=1}^p \delta_{iP(j)} \leq 1 \\ 1 & \text{if not,} \end{cases}$$

i.e., the switch  $s_i$  is "off" if the number of Ibu's belonging to processor  $p_i$  is one or less.

(2) Perform the following loop at most k times.

(a) Determine the processors  $l_{\text{max}}$  and  $l_{\text{min}}$  with

$$t^{l_{\max}} - t^{av} = \max\{t^1 - t_{av}, ..., t^k - t_{av}\}$$

and

$$t^{l_{\min}} - t^{\mathrm{av}} = \min\{t^1 - t_{\mathrm{av}}, ..., t^k - t_{\mathrm{av}}\}$$

If  $s_{l_{\text{max}}} = 0$  or  $s_{l_{\text{min}}} = 0$  then go to (3).

(b) Take the last line with index p in the list of lbu's of the processor  $l_{\text{max}}$ . If  $t_p \ge t^{l_{\text{max}}} - t^{l_{\text{min}}}$  then goto (3)

else E

move this lbu to the end of the list of lbu's of the processor  $l_{\min}$ and define  $t^{l_{\max}} := t^{l_{\max}} - t_p, t^{l_{\min}} := t^{l_{\min}} + t_p$ if  $t^{l_{\min}} < t_{av}$  and  $t^{l_{\max}} > t_{av}$  then goto (a)

- (c) if  $t^{l_{\max}} \le t_{av}$  then put  $s_{l_{\max}} = 0$ if  $t^{l_{\min}} \ge t_{av}$  then put  $s_{l_{\min}} = 0$ goto (a)
- (3) Stop the load balancing step.

*Remark* 3.3. This algorithm only uses the CPU-times of the set  $\{g_1, ..., g_p\}$  of lbu's to obtain an appropriate partition and the effort to perform this load balance step can be neglected in comparison with the global CPU-time.

Formally, the load balance step is not performed if the processors  $l_{\min}$  or  $l_{\max}$  of the present partition have one or less lbu's. In particular this is the case, if the partition has an idle processor. Assuming that in real applications the number of lbu's is large compared to the number of processors, one may even assume that the present partition does not include idle processors.

Numerical experiments show that this simple strategy yields a sufficient parallel efficiency as long as the communication between processors is sufficiently fast. We refer the reader to the numerical results presented in Section 4. Similar to the load balance techniques investigated in [14], the final partition turns out to be randomly distributed over the computational domain  $\Omega$ .

*Remark* 3.4. If the number of lbu's is of the same order as the number of processors, it might be useful to enlarge the Min–Max-Update described above using a binary exchange of lbu's along two processors.

## 3.2. The Streamline Approach

The Min–Max-Update described in the previous subsection yields a sufficiently accurate algorithm to obtain an appropriate load balance and works well if  $p/k \ge 1$  and the size of an lbu is sufficiently large as in the 3D-case. Nevertheless, in this approach there exists no strategy to involve an estimate on the communication between processors which is needed to handle the free flow condition in the splitting approach.

The idea to include some control on the communication is to arrange the subdomains, i.e., the lbu's belonging to one processor, along the streamlines of the flow. If the lbu's of one processor are located on a streamline, one may expect to obtain a low amount of communication because most of the particles will follow the streamlines.

Given a time-dependent velocity field  $u : \mathbb{R}_+ \times \Omega - \mathbb{R}^3$  on the spatial domain  $\Omega$ , a streamline is defined as a curve on  $\Omega$  such that the tangent is instantaneously everywhere parallel to the velocity field u. Especially, if the velocity field is time independent, the streamlines remain constant in time.

The Streamline Approach discussed in the following turns out to be an efficient load balance strategy for two-dimensional as well as axisymmetric computations where the computational domain has an inflow boundary on the  $x = x_{\min}$  boundary. Hence, we restrict ourselves to this situation: first we define for each lbu  $g_i$  the corresponding mean velocity  $v_i \in \mathbb{R}^2$  which is obtained by taking the arithmetic average over all particle velocities belonging to the lbu  $g_i$ . Moreover, we denote by  $g'_i, j = 1, ..., p^*$ , those lbu's which have an intersection with the inflow boundary of the domain  $\Omega$ . Then, the Streamline Approach to perform a partition updating P' is defined as follows: STREAMLINE APPROACH.

(1) Initialization.

put the inflow lbu's  $g'_j$ ,  $j = 1, ..., p^*$ , on a stack S and set  $s_i = 0$  for all i = 1, ..., p to indicate whether the lbu  $g_i$  already belongs to some processors.

put m = 1 and set  $t^i = 0$  for all i = 1, ..., k and assume that the averaged CPUtime  $t_{av}$  with respect to the present partition P is given.

(2) Following the Streamlines.

determine the lbu's belonging to processor  $p_m$  by performing the following steps

- (a) take lbu  $g'_{j_*}$  from top of the stack S, if the stack is empty go to (3)
- (b) If  $t_{j_*} + t_m \le t_{av}$  then put  $P'(j_*) = m$ ,  $s_{j_*} = 1$ ,  $t^m = t^m + t_{j_*}$ , and goto (b) else put m = m + 1, if m > k then go to (3) else go to (b) (a) consider the velocity vector  $y_{i_*} = (y_{i_*}^x, y_{i_*}^y)$
- (c) consider the velocity vector  $v_{j_*} = (v_{j_*}^x, v_{j_*}^y)$

if  $|v_{j_*}^x| > |v_{j_*}^y|$  then

choose the next lbu  $g_k$  in x-direction with respect to  $g_{j_*}$  if there exists no lbu go to (a)

```
if s_k = 0 then
```

```
put j_* = k and go to (b)
```

else

go to (a)

else

choose the next lbu  $g_k$  in y-direction with respect to  $g_{j_*}$ 

If there exists no lbu go to (a)

put the next lbu in x-direction with respect to  $g_{j_*}$  on top of the stack S if  $s_k = 0$  put  $j_* = k$  and go to (b) else go to (a)

(3) Distributing the rest.

determine the lbu's for which P(i) is undefined and put these lbu's on a stack S'

(a) determine the processor  $p_{l_{\min}}$  with minimal CPU-time  $t^{l_{\min}}$ 

(b) take lbu  $g''_{j_*}$  from top of the stack S' and put

$$P(j_*) = l_{\min} \quad t^{l_{\min}} = t^{l_{\min}} + t_{j_*}$$

if the stack is empty go to (4)

(c) If  $t^{l_{\min}} < t_{av}$  go to (b) else go to (a)

(4) Stop the load balancing step.

*Remark* 3.5. The decision 2(c) to proceed aligning the lbu's in the *x*- or *y*-direction approximates the technique of passing along a virtual streamline running through the lbu's. The reason to put the lbu located in the *x*-direction on top of the stack *S*, if one proceeds in the *y*-direction, is to allow the next processors to follow a streamline from this point on. The Streamline Approach is graphically illustrated in Fig. 1.

# 4. NUMERICAL RESULTS

In this section we present some rarefied gas simulations using various hardware platforms and compare the parallel efficiency of the adaptive load balance tech-



FIG. 1. Streamline approach in two dimensions.

niques described in the previous section. In particular we consider the following reentry configurations:

- 1. the EXPRESS capsule (Section 4.1)
- 2. a Deltawing configuration (Section 4.2)
- 3. the ARD capsule (Section 4.3).

The hardware platforms used to evaluate the parallel efficiency are given in Table II. The numerical results are obtained using the Finite-Pointset-Method (FPM) as described in [11, 15]. The implementation of this particle method on parallel computers (using the MPI communication library [9]) as well as the investigation of adaptive load balance techniques was done in the framework of the German HPCN-initiative ParBoSS [1].

In order to quantify the parallel efficiency of rarefied gas simulations the following notation is used in this section: The speedup factor using *n* processors is defined as the ratio of the total CPU-time  $T^n$  on *n* processors to the CPU-time  $T^1$  on a single node, i.e.,

$$Sp = \frac{T^1}{T^n}.$$

TABLE II

	Hardware	e Platforms		
System	Proc.	Туре	Mb	
INTEL Paragon	256	INTEL 860	32	
CRAY T3D	128	DEC Alpha 150 MHz	64	
nCUBE 2S	64	nCUBE 25 MHz	8	
IBM SP2	32	Power 2 66 MHz	128	
Pentium PC-cluster	8	Pentium 90 MHz	64	
HP-9000/K	2	PA-7200 100 MHz	256	



FIG. 2. EXPRESS capsule configuration.

Moreover, we define the parallel efficiency by

$$E = \frac{Sp}{n},$$

where n is the number of processors.

*Remark* 4.1. If the problem size is too large to run the simulation on a single node, we use an estimate on the single-node CPU-time. This estimate is obtained by summing up the CPU-times of the single processors without the communication time and without the additional effort to perform the load balancing steps. Tests against the exact single-node CPUs show that the estimated values for the speedup are even lower than the exact ones. For parallel computers with Cache memory, where one expects even a super-linear speedup factor, the single-node CPU-time is estimated using the minimum number of processors for which the simulation can be performed. Especially, the theoretic speedup values on these machines is in general underestimated.

The Load Balance Coefficient (lbc) is defined as the ratio between the maximal CPU-time over the single processors divided by the averaged CPU-time, i.e.,

$$lbc = \frac{t_{max}}{t_{av}},$$

where both CPU-times of the single processors are given as the total CPU-time minus the communication time and the time to perform the adaptive load balancing.

## 4.1. The EXPRESS Capsule

The EXPRESS capsule configuration (see Fig. 2) includes two pitot pressure probes: one at the (geometric) stagnation point as well as one along the side line.

	Parallel Efficien	cy with Static Lo	ad Balance		
Proc.	CPU [m:s]	Speedup	Effic.	lbc	
64	10:13	23.2	36.3	2.61	
32	11:15	21.1	65.9	1.48	
16	21:25	11.1	69.2	1.45	
8	35:20	6.7	83.9	1.20	
4	67:19	3.5	88.1	1.16	

TABLE IIIa Parallel Efficiency with Static Load Balance

The aim of the project is to measure atmospheric data during the reentry of the test configuration. The first calculations consider axisymmetric cases at zero angle of attack. The main purpose of rarefied gas simulations is the prediction of the flow conditions inside the pressure probe given the outer flowfield around the capsule. We refer the reader to the results given in [8].

Results on the parallel efficiency of a particle scheme for this configuration using an nCUBE 2S system with up to 64 nodes are shown in Tables IIIa–IIIc. The simulation uses about 60,000 particles to obtain the steady state solution after 800 time steps for a Knudsen number of 0.0961.

The spatial domain around the EXPRESS capsule is divided into  $60 \times 32 =$  1920 spatial cells and each cell represents one load balance unit. At time t = 0 each cell is filled with 28 particles which carry translational and rotational energy. The Mach number of the flow is equal to 27 with a freestream temperature of 192.6 K.

*Remark* 4.2. The adaptive load balance method, i.e., the partition updating of the lbu's, is performed in every 20th time step of the transient computation. One may even use an adaptive method here in order to reduce the number of load balance steps if the flow becomes stationary.

As shown in the tables, for low numbers of processors, the static load balance technique works reasonably well although the Min–Max-Update and the Streamline Approach already give better results. If the number of processors is increased, e.g., from 32 to 64 nodes, the parallel efficiency of the static partition drops dramatically down: the CPU-time using 32 and 64 nodes remains nearly constant. On the other hand, both adaptive load balance concepts still produce an increasing speedup

	Parallel Efficier	ncy Using Max–N	/lin-Update		
Proc.	CPU [m:s]	Speedup	Effic.	lbc	
64	7:19	32.4	50.7	1.67	
32	9:38	24.6	76.9	1.22	
16	17:26	13.6	85.0	1.14	
8	32:18	7.3	91.7	1.06	
4	60:08	3.9	98.6	1.007	

**TABLE IIIb** 

411

	Parallel Efficience	y Using Streamli	ne Approach		
Proc.	CPU [m:s]	Speedup	Effic.	lbc	
64	6:15	38.0	59.3	1.230	
32	8:46	27.1	84.6	1.035	
16	16:15	14.6	91.2	1.019	
8	31:17	7.6	94.7	1.012	
4	60:59	3.9	97.2	1.010	

 TABLE IIIc

 Parallel Efficiency Using Streamline Approach

factor, where the Streamline Approach gives better results and the difference to the Min–Max-Update increases with the number of processors. Similar results using an adaptive load balance technique are obtained using different parallel systems: Table IV shows some results using the Streamline Approach as adaptive load balance technique.

System	Proc.	CPU [m:s]	Speedup	Effic.	lbc
CRAY T3D	64	1:50	36.5	57.0	1.42
	32	2:29	27.0	84.3	1.05
	16	4:39	14.0	87.3	1.02
	8	9:02	7.4	92.6	1.04
	4	17:39	3.8	94.8	1.03
	2	34:01	1.97	98.4	1.008
	1	66:57	-	-	-
nCUBE 2S	64	6:15	38.0	59.3	1.230
	32	8:46	27.1	84.6	1.035
	16	16:15	14.6	91.2	1.019
	8	31:17	7.6	94.7	1.012
	4	60:59	3.9	97.2	1.010
IBM SP2	32	1:23	19.8	61.7	1.13
	16	2:06	13.0	81.1	1.03
	8	3:59	6.9	85.7	1.008
	4	7:10	3.8	95.4	1.004
	2	13:52	1.97	98.6	1.004
	1	27:20	-	_	-
PC	8	8:10	6.65	83.1	1.03
	4	14:05	3.86	96.4	1.02
	2	27:07	2.003	100.15	1.005
	1	54:19	-	-	-
HP	2	17:33	1.88	94.2	1.009
	1	33:04	-	-	-

TABLE IV Parallel Efficiency Using Streamline Approach

	Parallel	Efficiency Using	vin-wax-Opdate	e	
System	Proc.	CPU [m:s]	Speedup	Effic.	lbc
CRAY T3D	128	17:49	120.6	94.2	1.05
	64	34:10	62.9	98.3	1.02
	32	68:55	31.2	97.4	1.009
	16	136:39	15.7	98.3	1.005
nCUBE 2S	64	139:37	60.8	94.9	1.02
	32	273:48	31.0	96.8	1.01
IBM SP2	32	34:40	32.3	100.9	1.01
	16	68:53	16.3	101.6	1.007
	8	138:50	8.1	100.8	1.005
	4	282:38	3.96	99.0	1.004

 TABLE V

 Parallel Efficiency Using Min–Max-Update

The speedup factors given in Table IV show that, although applying an adaptive load balance technique, the parallel efficiency drops down when the number of processors is increased. This indicates that the problem size becomes too small using a large number of nodes, i.e., the problem size must fit the computational power to obtain a reasonable parallel efficiency.

As an example, one may pass to a full three-dimensional computation using the same mesh size. Introducing a further spatial dimension increases the problem size by one order of magnitude. The results on the parallel efficiency for the three-dimensional calculation using the Min–Max-Update are given in Table V.

The parallel efficiency remains reasonably high even if the number of processors is increased.

*Remark* 4.3. Numerical experiments show that the Streamline Approach gives less efficient results for three-dimensional computations. This is due to the fact that the approximation of real streamlines by the technique given in Section 3.2 is not accurate enough. Hence, one loses the advantage of a lower communication effort and, at the same time, the computational effort of applying the Streamline Approach is much higher than in the Min–Max-Update.

## 4.2. Deltawing Configuration

Hypersonic rarefied flow simulations around a deltawing configuration are classical test problems for rarefied gas simulation techniques. Several authors already investigated this configuration using various simulation tools [5, 6]. The present configuration is shown in Fig. 3.

Although one expects a high communication between the single processors, the Min–Max-Update gives a quite reasonable parallel efficiency. Table VI gives some results using various hardware platforms as well as various numbers of processors. The simulation is performed at a Knudsen number of 0.1 and a Mach number of



FIG. 3. Deltawing configuration.

8.9 on a  $20 \times 64 \times 20$  grid and about 1,000,000 particles using 300 time steps. One lbu consists of 80 cells, i.e., the spatial domain consists of 320 lbu's. Moreover, the load balance steps are performed at every 10th time step of the transient computation.

	1 urune	Emelency esing	in in the option	···	
System	Proc.	CPU [m:s]	Speedup	Effic.	lbc
Paragon	256	2:06	182.0	71.1	1.15
	128	3:18	115.5	90.3	1.08
	64	6:09	62.2	97.11	1.03
	32	12:39	30.2	94.4	1.02
CRAY T3D	128	1:28	115.6	90.3	1.06
	64	1:50	61.4	95.9	1.02
	32	2:29	31.2	97.5	1.01
	16	4:39	15.8	98.9	1.007
	8	9:02	7.9	99.0	1.005
	4	42:54	3.9	98.8	1.003
nCUBE 2S	64	10:54	57.2	89.3	1.02
	32	20:46	30.0	93.7	1.02
	16	40:24	15.4	96.4	1.007
IBM SP2	32	3:04	33.9	106.1	1.02
	16	5:47	18.0	112.4	1.01
	8	11:50	8.8	109.8	1.005
	4	24:41	4.2	105.5	1.005
	2	50:33	2.06	103.0	1.002
	1	103:54	-	_	-
Pentium	8	24:41	7.38	92.3	1.02
	4	47:45	3.82	95.4	1.007
HP	2	56:36	1.99	99.5	1.002
	1	112:41	-	-	-

TABLE VI Parallel Efficiency Using Min–Max-Update



FIG. 4. ARD capsule configuration.

Two special points have to be mentioned

• The IBM SP2 system gives a speedup which is for all nodes greater than the theoretical value. This is due to the increasing efficiency of the cache system.

• The INTEL Paragon drops dramatically going from 128 to 256 nodes. This indicates that the current problem is too small to use a large number of nodes.

#### 4.3. ARD Capsule

The ARD capsule configuration (see Fig. 4) was investigated in the framework of the ESA-project "Development of Prediction Tools for Re-entry Capsule Aero-thermodynamics." The aim of the project is to study the influence of chemical reactions on the aerothermodynamic characteristics of reentry configurations. The current simulations use a 5 species air model consisting of  $N_2$ ,  $O_2$ , N, O, and NO together with 19 dissociation and exchange reactions where the rate coefficients are based on the Dunn–Kang model (recombination reactions are neglected).

The modelling of the chemical cross sections and the implementation of chemical reactions in a particle scheme are based on the work given in [13, 16]. Numerical results are obtained for two altitudes at 94 and 103 km. For a detailed description and more information concerning the physical relevance of the results we refer the reader to [16].

As already mentioned, the aim of the simulations presented in the following is to study the influence of chemical reactions in rarefied gas flows. One expects that chemical reactions become more important as the altitude of the reentry configuration decreases because of the higher collision frequency at lower altitudes. Hence, two computations are performed for both altitudes at 94 and 103 km. The first simulation uses a gas mixture consisting of nitrogen and oxygen molecules with mole fractions of 0.756 and 0.244. The second simulation includes the formation of nitrogen and oxygen atoms as well as nitric oxide due to chemical reactions (dissociation and exchange reactions).

	Physical	Parameters fo	or the ARD-Ca	psule	
H [km]	$\rho_{\infty}  [kg/m^3]$	$\lambda_{\infty}$ [m]	Kn	$V_{\infty}$ [m/s]	$T_{\infty}$ [K]
94	$1.459 \cdot 10^{-6}$	0.05	0.0178	7768	193
103	$2.945 \cdot 10^{-7}$	0.26	0.0929	7758	224

TABLE VIIPhysical Parameters for the ARD-Capsule

*Note.* H is the altitude of the reentry configuration;  $\rho_{\infty}$  is the freestream density;  $\lambda_{\infty}$  is the freestream mean free path; Kn is the Knudsen number;  $V_{\infty}$  is the freestream velocity; and  $T_{\infty}$  is the freestream temperature.

All simulations include an angle of attack of 20 degree such that the full threedimensional ARD-configuration has to be used, although the ARD-capsule is defined as an axisymmetric configuration. The physical parameters used for both altitudes are summarized in Table VII. Moreover, the boundary condition at the surface of the ARD-capsule is given by complete accommodation with diffusive reflection at a wall temperature of 600 K for the translational and rotational energies. The vibrational energies remain unchanged in the gas–surface interaction process. The corresponding numerical parameters and the parallel efficiency using different architectures are given in Table VIII. In all simulations the Min–Max-Update is applied to obtain a reasonable performance on parallel computers and the load balance steps are performed at every 20th time step.

Moreover, the simulation includes an adaptive grid refinement in order to cover the large density deviations over the spatial domain. In all simulations a reasonable parallel efficiency is achieved using the Min–Max-Update. Finally, the influence of chemical reactions on the aerothermodynamic characteristics of the ARD-configuration is given in Table IX.

#### **5. CONCLUSION**

In order to achieve a reasonable parallel efficiency of rarefied gas simulations using particle methods it is necessary to introduce appropriate adaptive load balance techniques. Due to the high dimensionality of the underlying optimization problem, it is moreover necessary to develop heuristic methods. Two such approaches are

			IAD						
Numerical Parameters for the ARD-Capsule									
Н	Chem.	Hardware	Proc.	CPU	Speedup	Effic.	N <sub>part</sub>		
94	No	Cray T3D	128	4:39	118.1	92.3	$9.2 \cdot 10^{6}$		
94	Yes	Cray T3D	128	8:58	119.6	93.4	$11.3 \cdot 10^{6}$		
103	No	nCUBE 2S	64	6:43	55.9	87.3	$1.6 \cdot 10^{6}$		
103	Yes	Cray T3D	32	3:16	30.8	93.2	$1.7 \cdot 10^{6}$		

TARLE VIII

*Note.* H is the altitude of the reentry configuration; CPU is the total CPU-time in hours and minutes; and N<sub>part</sub> is the total number of simulation particles.

	Aeromermodynamic Characteristics of the ARD-Capsule								
H [km]	Chem.	CA	CN	Cm	CD	CL	Ch		
94	No	1.432	0.169	-0.079	1.403	-0.330	0.210		
94	Yes	1.439	0.170	-0.080	1.411	-0.332	0.160		
103	No	1.535	0.253	-0.088	1.529	-0.286	0.400		
103	Yes	1.535	0.254	-0.089	1.530	-0.286	0.382		
94 103 103	Yes No Yes	1.439 1.535 1.535	0.170 0.253 0.254	-0.080 -0.088 -0.089	1.411 1.529 1.530	-0.332 -0.286 -0.286	0.1 0.4 0.3		

TABLE IX
Aerothermodynamic Characteristics of the ARD-Capsule

*Note.* CA is the axial force coefficient; CN is the normal force coefficient; Cm is the pitching moment based on the stagnation point; CD is the drag coefficient; CL is the lift coefficient; and Ch is the heat transfer coefficient.

presented in the current investigation. Introducing the concept of load balance units (lbu's) one might use a simple Min–Max-Update between the single processors based on the CPU-times as well as a more sophisticated method where one tries to align the load balance units along streamlines (Streamline Method).

Numerical experiments show that the latter one yields a reasonable load balance for axisymmetric (as well as two-dimensional) simulations. Using an even simpler technique, like the Min–Max-Update, will actually increase the parallel efficiency in three-dimensional computations. Moreover, the results obtained are nearly independent of the underlying hardware platforms.

The numerical results presented in the paper include three different reentry configurations as well as a wide range of different parallel systems and the results clearly indicate the efficiency of rarefied gas simulations on parallel computers when using an adaptive load balance strategy. Nevertheless, the results also show the well-known fact that the efficiency is correlated with the size of the investigated problem. The results presented in the paper use adaptive load balance steps after a fixed number of time steps. Techniques concerning how to include an adaptive method to perform the partition updating in order to reduce the number of load balance steps if the flow becomes stationary are currently under investigation.

# ACKNOWLEDGMENTS

The main part of this research was funded by the German Minister for Research and Technologies under Contracts 01-IR-408-A-0 and 03-NE7KAI. Moreover, the research was supported by the European Space Agency under Contract 11171/94/F/WE, "Development of Prediction Tools for Re-entry Capsule Aerothermodynamics" (ARD capsule).

#### REFERENCES

- S. Antonov, M. Hartig, F. J. Pfreundt, W. Sack, K. Steiner, J. Struckmeier, A. Schüller, and G. Koppenwallner, BMBF-Verbundprojekt ParBoSS: Mathematik-Parallelisierung-Anwendungen, in *Statustagung des BMBF HPSC 95*, edited by G. Wolf and R. Krahl (DLR Report, Berlin, 1995), p. 67.
- S. Antonov, K. Steiner, and J. Struckmeier, Development of Prediction Tools for Re-entry Capsule Aerothermodynamic, Final Report, ESA Contract 11171/94/F/WE, 1996.

- L. Dagum, Comparisons of two approaches to direct particle simulation on the connection machine, in *Proceedings, 18th Int. Symp. on Rarefied Gas Dynamics*, edited by B. D. Shizgal and D. P. Weaver (AIAA, Washington, 1994), p. 429.
- S. Dietrich and I. D. Boyd, Scalar and parallel optimized implementation of the direct simulation Monte Carlo method, J. Comput. Phys. 126, 328 (1996).
- F. Gropengießer, H. Neunzert, J. Struckmeier, and B. Wiesen, Rarefied gas flow around a 3Ddeltawing, in *Hypersonic Flows for Reentry Problems*, Vol. II, edited by J. A. Desideri *et al.* (Springer-Verlag, Heidelberg, 1991), p. 1067.
- F. Gropengießer, H. Neunzert, J. Struckmeier, and B. Wiesen, Rarefied gas flow around a 3D delta wing, in *Hypersonic Flows for Reentry Problems*, Vol. III, edited by R. Abgrall, *et al.* (Springer-Verlag, Heidelberg, 1992), p. 976.
- 7. G. A. Bird, *Molecular Gas Dynamics and the Direct Simulation of Gas Flows* (Clarendon, Oxford, 1994).
- A. Hartmaier, H. M. Urbassek, and W. Sack, Transition flow in an impact pressure probe: Relevance of gas-wall interaction, *Phys. Fluids* 9, 1344 (1997).
- 9. MPI, Message-passing interface standard, in Message Passing Interface Forum, May 1994.
- 10. P. P. Nance, R. G. Wilmoth, B. Moon, H. A. Hassan, and J. Saltz, Parallel Monte Carlo simulation of three dimensional flow over a flat plate, *J. Thermophys. Heat Transfer* **9**, No. 3, 471 (1995).
- 11. H. Neunzert and J. Struckmeier, Particle methods for the Boltzmann equation, in *Acta Numerica* 1995 (Cambridge Univ. Press, Cambridge, UK, 1995), p. 417.
- C. D. Robinson and J. K. Harvey, A parallel DSMC implementation on unstructured meshes with adaptive domain decomposition, preprint, presented at the 20th RGD Symposium, Beijing, China, 1996.
- 13. W. Sack, *Modelling and Numerical Simulations of Chemically Reacting Flows in Rarefied Gases*, Ph.D. thesis (University Kaiserslautern, 1995).
- 14. J. Struckmeier and F. J. Pfreundt, On the efficiency of simulation methods for the Boltzmann equation on parallel computers, *Parallel Comput.* **19**, 103 (1993).
- 15. J. Struckmeier and K. Steiner, A comparison of simulation methods for rarefied gas flows, *Phys. Fluids A* **7**, 2876 (1995).
- K. Steiner and J. Struckmeier, Modelling and simulation of chemically reacting rarefied gas flows, in preparation.
- R. G. Wilmoth, Application of a parallel DSMC method to hypersonic rarefied flows, AIAA 91-0772, 1991.